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Filter Design for Failure Detection and Isolation in the Presence of Modeling Errors and Disturbances*

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Abstract

The design problem of filters for robust Failure Detection and Isolation, (FDI) is addressed in this paper. The failure detection problem will be considered with respect to both modeling errors and disturbances. Both an approach based on failure detection observers as well as an approach based on a standard setup optimization is presented in this paper.

1 Introduction

The main problem in failure detections for dynamic control systems is to distinguish failures from other disturbances, facing model uncertainties in the system. In the past 10 - 15 years, a lot of papers has been considering the failure detection problem. Good survey papers can be found in [11] and in [4]. Further, the book by Patton et al. [10], also gives a good introduction to the area of failure detection.

In the past years, the robustness aspect has also been introduced in relation with failure detection as e.g. in [12], [1], [13], [3] and in [9] to mention a few papers.

The FDI methods described in this paper are related to other methods derived in this area which are either based on observer techniques or on an optimization approach, like \mathcal{H}_∞ or \mathcal{L}_1 or combinations of these. These two types are not the only types of methods for failure detections and isolations. Another class of methods is based on statistical approaches, see [7].

If we consider filters for FDI based on optimizations design, a lot of different methods has been derived. A general observer based approach in [5] has been

applied in connection with an \mathcal{H}_∞ optimization. A result of these methods can be found in [6], where the benchmark problem from [2] has been considered. A direct formulation of the FDI problem as a robust \mathcal{H}_∞ filter design problem has been given in [3]. This approach results in solving a standard Riccati equation for a standard \mathcal{H}_∞ filter problem. Another \mathcal{H}_∞ approach to the FDI problem can be found in [12]. An approach based on \mathcal{H}_∞ optimization is given in [9], where the robust FDI design is combined with statistical methods for the detection of failure signals. The result given in the paper is very convincing.

The reason for using \mathcal{H}_∞ optimization in connection with the FDI problem is due to the robustness aspect with respect to model uncertainties. However, from a FDI point of view, it is more natural to use an \mathcal{L}_1 optimization of the FDI filter. By using an \mathcal{L}_1 optimization of the FDI filter, we will obtain that the magnitude of the residual signal with respect to disturbances and model uncertainties are bounded in magnitude. This will make the detection of failure signals more readily comparable to the case where an \mathcal{H}_∞ optimization has been applied. In this case, the residual signal is bounded in the power semi-norm. An \mathcal{L}_1 optimization for integrated control and FDI has been applied in [1].

The rest of this paper is organized as follows. First, in Section 2 we describe an observer based solution. This setup is quite intuitive. Based on this setup, we formulate the robust failure detection problem in a standard problem framework in Section 3, and we provide several approaches to its solution based on optimization methods. Section 4 summarizes the results.

2 An Observer Based FDI Approach

We shall follow the observer based approach derived by Patton et al. described in e.g. [10], [11] or in [8]. Together with the observer, an output weight matrix is included as an additional design parameter. In this section we shall demonstrate how to modify

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the observer structure in order to allow for systematic design of the weight matrix and the observer simultaneously by formulating it as a standard robust design problem. In contrast, using a standard setup, the design of the observer and the weight matrix would be coupled.

Consider the following system G given by:

$$\begin{aligned}\dot{x} &= Ax + B_1w + B_2d + B_3f \\ y &= Cx + D_1w + D_2d + D_3f\end{aligned}\quad (1)$$

(We shall throughout the paper assume 'compatible' dimensions of vectors and matrices to avoid tedious listings of dimensions.)

Now, let's use the failure detection approach described in e.g. [11] or in [8], where an observer together with a weight matrix are applied for the failure detection. Let the full order observer be given by:

$$\dot{\hat{x}} = A\hat{x} + K(y - C\hat{x}) \quad (2)$$

where K is the observer gain and \hat{x} is the state estimate. The output error e_y is given by

$$e_y = y - C\hat{x}$$

By using the observer on the system in (1), we get the following equation for the output error e_y

$$\begin{aligned}e_y &= C(sI - A - KC)^{-1}(B_1 + KD_1)w \\ &\quad + C(sI - A - KC)^{-1}(B_2 + KD_2)d \\ &\quad + C(sI - A - KC)^{-1}(B_3 + KD_3)f \\ &\quad + D_1w + D_2d + D_3f \\ &= G_{ew}(K, s)w + G_{ed}(K, s)d + G_{ef}(K, s)f\end{aligned}\quad (3)$$

To realize a failure detection and isolation scheme, we will premultiply the output error e_y by a prefilter $H(s)$, to generate the residual:

$$\begin{aligned}r(s) &= H(s)G_{ew}(K, s)w + H(s)G_{ed}(K, s)d \\ &\quad + H(s)G_{ef}(K, s)f\end{aligned}\quad (4)$$

The design parameters are then the observer gain K and the filter matrix H . The design of these matrix gains are done separately in the design method applied in [11], [8]. The observer gain is designed by an Eigenstructure Assignment method and the following design of H is more ad hoc. However, here we want to derive a combined design method for the observer gain and the prefilter by formulating the design problem in a standard setup. First, let's look at the design conditions we have to the transfer matrices from w , d and f to r in (4). It is clear, that

we want to minimize the transfer matrix from the external input w to r and also the transfer function from d to r . These transfer function matrices represent model uncertainties and disturbances, which we want to make the residual robust against. The transfer matrix from the failure signals f to the residual r must not be small, due to the fact that we want to detect the failure signals in the residual vector. Therefore, we can instead look at the transfer matrix given by $I - HG_{ef}$. By minimizing this transfer matrix, we will obtain that the residual vector gives a good estimate of the failure vector. To summarize, we have the following design conditions for the design problem:

Problem 1 Let the residual vector be given by

$$r(s) = HG_{ew}(K, s)w + HG_{ed}(K, s)d + HG_{ef}(K, s)f$$

where the controller parameters are K and H . Design K and H such that

$$HG_{ew} \approx 0$$

$$HG_{ed} \approx 0$$

$$HG_{ef} \approx I$$

The conditions will not be possible to meet uniformly in frequency, but should be met at "appropriate frequencies". We will elaborate on this below.

After a little matrix manipulation, the transfer matrix from w , d and f to r can be described as an LFT in the controller parameters K and H , i.e. $\mathcal{F}_l(\tilde{G}(s), \tilde{K}(s))$, where $\tilde{G}(s)$ is given by

$$\tilde{G}(s) = \left[\begin{array}{c|ccc|cc} A & B_1 & B_2 & B_3 & [I & 0] \\ \hline 0 & 0 & 0 & -I & [0 & I] \\ C & D_1 & D_2 & D_3 & [0 & 0] \end{array} \right] \quad (5)$$

and the controller $\tilde{K}(s)$ is given by

$$\tilde{K}(s) = \begin{bmatrix} K \\ H \end{bmatrix} \quad (6)$$

In the original design setup by Patton et al. [11], the two gains K and H are constant gains, but this will not in general be the case now. Here we will get two dynamic controllers which will satisfy the design conditions. By just using constant gains, we will not in general get an optimal observer/filter.

The structure described here is depicted in Figure 1.

The $\Delta(s)$ block represent the robust stability condition. The $\Delta_f(s)$ and $\Delta_d(s)$ blocks represent performance specifications, i.e. specifications for the disturbance attenuation and for the performance specification for the failure signals.

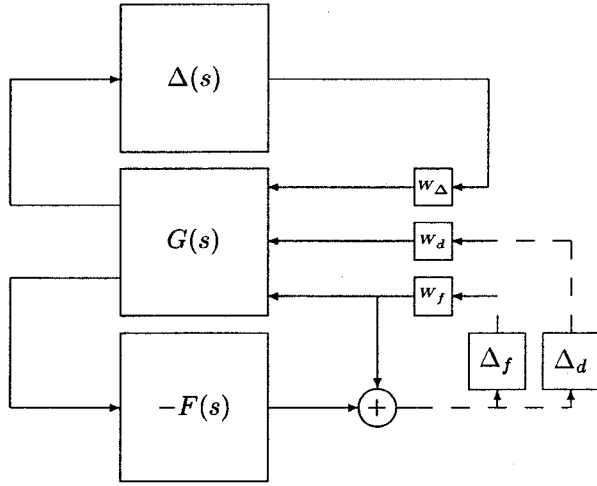


Figure 3: A filter based robust FDI structure

3 Standard Problem Formulation

The approach in this section is appealing in the sense that it is based on a very straightforward and precise problem formulation for the robust FDI problem which can then be solved in one step. Based on the setup in Figure 3, we will derive the setup directly. The structure of the controller will not, though, be intuitive in any sense as in the case where an observer is applied. We consider the following model for a (potentially) faulty uncertain dynamical system:

$$\begin{aligned} \dot{x} &= Ax + B_1w + B_2d + B_3f \\ z &= C_1x + D_{11}w + D_{12}d + D_{13}f \\ y &= C_2x + D_{21}w + D_{22}d + D_{23}f \end{aligned} \quad (8)$$

x is the state vector. w and z describe the uncertainty dynamics, i.e., the uncertainty is described as a feedback interconnection from z to w with a feedback of the form $W_\Delta \Delta$, $\|\Delta\|_\infty < 1$. W_Δ describes the frequency dependence of the uncertainty. This setup is standard and fairly general.

d is the vector of external disturbances and noises. By selecting matrices B_2 and D_{22} carefully, it is possible to model both the exogenous signals that enter the dynamics as well as those that influence the measurements only.

f is the failure vector, which include both sensor and actuator failures (by selection of the B_3 and D_{23} matrices).

Finally, y is the measurement output.

We shall now set up an optimization problem in standard form. In order to obtain useful results it is of paramount importance to introduce appropriate weightings. In addition to the uncertainty model W_Δ which will typically possess high pass properties, we need weightings that describe the disturbance and

fault models. Hence, we assume the signal d to be generated by a frequency unbiased signal w_2 passing through a filter W_d :

$$d = W_d w_2$$

Typically, W_d will be of low pass character if it describes process disturbances and of high pass character if it is a sensor noise model.

Similarly, we assume f to be generated by a signal w_3 with flat frequency characteristics:

$$f = W_f w_3$$

Typically W_3 will have a low pass character. If only static failure signals are anticipated (as will often be the case) it is important to assign the roll off frequency for this filter much lower than the system cross over frequency.

The objective of our optimization is to generate a signal u which approximates the faults z as well as possible. In other words, we wish the following error:

$$z_2 = u - f$$

to be as small as possible, subject to the modeling error Δ , and in presence of the exogenous signals d .

Assuming state space models of the three weights as follows:

$$W_i = \left[\begin{array}{c|c} A_i & B_i \\ \hline C_i & D_i \end{array} \right], \text{ for } i = \Delta, d, f$$

the corresponding standard model, takes the following form:

$$\bar{G} = \left[\begin{array}{c|cccc} \bar{A} & \bar{B}_1 & \bar{B}_2 & \bar{B}_3 & \bar{B}_4 \\ \hline \bar{C}_1 & \bar{D}_{11} & \bar{D}_{12} & \bar{D}_{13} & \bar{D}_{14} \\ \bar{C}_2 & \bar{D}_{21} & \bar{D}_{22} & \bar{D}_{23} & \bar{D}_{24} \\ \bar{C}_3 & \bar{D}_{31} & \bar{D}_{32} & \bar{D}_{33} & \bar{D}_{34} \\ \bar{C}_4 & \bar{D}_{41} & \bar{D}_{42} & \bar{D}_{43} & \bar{D}_{44} \end{array} \right] \quad (9)$$

where the inputs are w_1, w_2, w_3 and u , the outputs are z_1, z_2, z_3 and y . The involved matrices are as follows:

$$\bar{A} = \left[\begin{array}{cccc} A & B_1 C_\Delta & B_2 C_d & B_3 C_f \\ 0 & A_\Delta & 0 & 0 \\ 0 & 0 & A_d & 0 \\ 0 & 0 & 0 & A_f \end{array} \right]$$

$$\begin{bmatrix} \bar{B}_1 & \bar{B}_2 & \bar{B}_3 & \bar{B}_4 \end{bmatrix} = \left[\begin{array}{cccc} B_1 D_\Delta & B_2 D_d & B_3 D_f & 0 \\ B_\Delta & 0 & 0 & 0 \\ 0 & B_d & 0 & 0 \\ 0 & 0 & B_f & 0 \end{array} \right]$$

$$\begin{aligned}\bar{C}_1 &= [C_1 \quad D_{11}C_\Delta \quad D_{12}D_d \quad D_{13}D_f] \\ \bar{D}_1 &= [D_{11}D_\Delta \quad D_{12}D_d \quad D_{13}D_f \quad 0] \\ \bar{C}_2 &= \begin{bmatrix} 0 \\ 0 \\ 0 \\ -C_f \end{bmatrix}, \quad \bar{D}_2 = [0 \quad 0 \quad -D_f \quad I] \\ \bar{C}_3 &= \bar{C}_2, \quad \bar{D}_3 = \bar{D}_2. \\ \bar{C}_4 &= [C_2 \quad D_{21}C_\Delta \quad D_{22}D_d \quad D_{23}D_f] \\ \bar{D}_4 &= [D_{21}D_\Delta \quad D_{22}D_d \quad D_{23}D_f \quad 0]\end{aligned}$$

Note that z_3 is just a copy of z_2 , thus making the system square (as required by some optimization procedures).

So far we have not been specific concerning the method of optimization to be employed for the standard problem (9).

Ideally speaking, the choice of method for optimization should be governed by the available data for the specific physical plant. E.g. if the disturbance model has been identified using bounded power amplification an \mathcal{H}_∞ setup should be used, and if was obtained using time domain methods \mathcal{L}_1 methods would be the right choice.

However, very few methods are actually available that would effectively compute a useful filter for our problem. The reason is mainly that most robust methods are based on a robust stability/nominal performance principle. This is not appropriate for the problem we are considering, which is indeed an intricate one, because we are considering so many different effects at the same time (dynamical uncertainty, disturbances, noises, and faults). The only way to distinguish dynamical uncertainty from the other effects is to pose the problem as a *robust performance* problem. Robust stability/nominal performance methods would treat w_1 simply as an exogenous signal just as w_2 and w_3 , and this approach would work only in very simple cases. Moreover, the disturbance model corresponding to (9) is based on structured uncertainty, which the method should also be able to handle.

The current state of art mainly offers two methods for design of robust performance in the face of structured uncertainty. These are \mathcal{L}_1 and μ optimization methods. In our opinion, these methods would be the best choices for small and medium size systems.

The following result describes the resulting behavior of a filter $F(s)$ designed by μ synthesis.

With some lack of stringence we will call the essential supremum of μ over all frequencies the μ norm. By the \mathcal{L}_1 norm we shall mean the operator norm of an operator mapping \mathcal{L}_∞ signals (amplitude bounded) into \mathcal{L}_∞ signals.

Theorem 1 Let $F(s)$ be a given transfer function. The following two are equivalent:

1. Applying the feedback $u = F(s)y$ for the system (9) stabilizes that system and makes the μ norm (the \mathcal{L}_1 norm) smaller than one with respect to the structure:

$$\bar{\Delta} = \begin{bmatrix} \Delta_w & 0 & 0 \\ 0 & \Delta_d & 0 \\ 0 & 0 & \Delta_f \end{bmatrix}$$

2. Applying the filter $F(s)$ to the output of the original system (8) generates a signal u with the property that the error signal $u - f$ has power bounded by 1, provided*

- (a) the \mathcal{H}_∞ norm of the uncertainty weighted by the inverse of W_Δ is bounded by 1
- (b) the power of d weighted by the inverse of W_d is bounded by 1
- (c) the power of f weighted by the inverse of W_f is bounded by 1

A similar result holds for \mathcal{L}_1 optimization, simply by replacing the norms in the above theorem (and formulating it in time domain rather than frequency domain).

The multi objective method mentioned above will provide necessary conditions only for robust performance, but the interpretation will be the same as for the μ formulation.

The design algorithm we are putting forward is based on controller optimization techniques, although we are using these techniques to design a filter rather than a controller.

Algorithm 1 Indata: system matrices as in (9). Outdata: Filter $F(s)$.

1. Formulate weightings such that the faults would be isolated by the power signal interpretations in Theorem 1 (or by the corresponding time domain properties for \mathcal{L}_1 optimization)
2. Compute the matrices in order to set up the standard problem (9)
3. Apply μ synthesis (or \mathcal{L}_1 synthesis) to the problem (9)
4. The resulting controller $F(s)$ is the robust FDI filter for the system (8)

*The formulation of the last three conditions is not entirely stringent, since the inverses need not be neither stable nor proper. This slightly sloppy formulation is chosen to provide the intuition for the power signal interpretation, which would be relevant to an FDI application.

4 Conclusion

The failure detection and isolation in presence of modeling errors and disturbance problem has been considered in this paper by using two different approaches.

The first approach applied to the FDI problem is based on using observers together with a prefilter. This design approach has been formulated as a robust standard design problem. In previous approaches in the literature, the observer and the prefilter was designed separately, which does not guarantee optimality. If instead we use the robust standard formulation, the design of observer and prefilter is done in one step. However, for making an optimal filter for the FDI problem, we need only the prefilter. So, as the result of this approach, the observer based approach is equivalent with a standard filter design in the optimal case, but it offers additional information which might be desirable in itself.

The other approach is based on a direct formulation of the FDI design problem, and offers a completely optimization based design of an unstructured filter, which need not be observer based in any intuitive way.

Both approaches offer implementation under several optimization schemes which might be given by the application. It is argued that from current state-of-the-art methodologies, μ or \mathcal{L}_1 based methods would yield the best results for FDI applications, such as the one addressed in this paper.

Depending on how large the design problem is (number of states, number of inputs and outputs etc.), it might be more convenient to use a simpler multi objective design approach based on sensitivity optimization, rather than \mathcal{H}_∞ , μ , or \mathcal{L}_1 methods for computational reasons.

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